## **TECHNICAL NOTE**

# COMPUTER AIDED ANALYSIS OF REINFORCED CONCRETE COLUMNS SUBJECTED TO AXIAL COMPRESSION AND BENDING—I L-SHAPED SECTIONS

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Abstract—Numerical investigations on the strength of L-shaped short reinforced concrete columns subjected to combined axial load and bending were undertaken for the purpose of providing design aids for structural engineers. The use of a computer lends itself naturally to the solution of the problem which generally requires an iterative process. Therefore, an attempt has been made in this paper to computerize the analysis procedure for L-shaped sections and in the accompanying paper (part II)<sup>2</sup> for T-shaped column sections. The ACI-318, CP-110 and IS-456 codes presented design aids only for square/rectangular and circular columns. Apparently this study constitutes the first to present the interaction curves for L-shaped and T-shaped column sections with the limit state analysis.

## INTRODUCTION

The analysis and design of L-shaped corner columns are complicated and cumbersome. Next to rectangular and circular shapes, L-sections may be the most frequently encountered reinforced concrete columns, since they can be used at outside and re-entrant building corners. Nevertheless, information for their analysis and design is not generally available to structural engineers, either in working stress or ultimate strength theories. There are some design approaches in which the design effort is reduced by approximated shape of strength envelopes (e.g. Bresler [2], Parme et al. [3], CP-110 [4], ACI-318 [5] and IS-456 [6, 7]), and the use of simplifying approximations (e.g. the method of superposition [8] and the method of equivalent uniaxial eccentricity [8]). Ramamurthy [9] developed simple equations to closely represent the load contours in square and rectangular columns. He also illustrated how they can be used to determine the appropriate interaction diagram for given eccentricities of the load. Although several noteworthy articles [2, 3, 8-17] on biaxial bending of square/rectangular column sections, which contributed greatly to the understanding of this subject, have appeared in recent years, significant gaps in the area of design aids for biaxial bending still exist. To lessen these gaps, a number of comprehensive design aids are presented in the present investigation.

This paper deals with the limit state analysis of L-shaped reinforced concrete (R.C.) columns. The aim of limit state design is to achieve acceptable probabilities that the structure will not become unfit for the use for which it is intended, that is, that it will not reach a limit state. To ensure the above objectives, the design should be based on characteristic values for material strengths and applied loads, which takes into account the variations in the material strengths and in the loads to be supported.

## ASSUMPTIONS AND MATERIAL PROPERTIES

In the analysis, the following assumptions [6], which are almost the same as those codified in CP-110 [4], are made:

- (a) the strain distribution in the concrete in compression and the strain in the reinforcement, whether in tension or compression, are derived from the assumption that plane sections normal to the axis remain plane after bending, and that there is no bond-slip between the reinforcement and the concrete,
- (b) the tensile strength of concrete is ignored,
- (c) the relationship between stress-strain distribution in concrete is assumed to be parabolic as shown in Fig. 1. The maximum compressive stress is equal to  $0.67 f_{ck}/1.5$  (see Fig. 2),
- (d) the stresses in reinforcement are derived from the representative stress-strain curve for the type of steel used. Typical curves are shown in Figs 3 and 4,
- (e) the maximum compressive strain in concrete in axial compression is taken as 0.002,
- (f) the maximum compression strain at the highly compressed extreme fibre in concrete subjected to axial compression and bending, but when there is no tension on the section, is taken as 0.0035 minus 0.75 times the strain at the least compressed extreme fibre (see Fig. 2a),
- (g) the maximum compressive strain at the highly compressed extreme fibre in concrete subjected to axial compression and bending, when part of the section is m tension, is taken as 0.0035 (see Fig. 2b). In the limiting case, when the neutral axis lies along one edge of the section, the strain varies from 0.0035 at the highly compressed edge to zero at the opposite edge.

### METHOD OF ANALYSIS

The criteria generally proposed for determining the ultimate strength of R.C. members subjected to axial compression combined with bending are based on limiting the maximum strain (or stress) in the concrete to some prescribed value. The load-carrying capacities discussed

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<sup>‡</sup> See ref. [1].

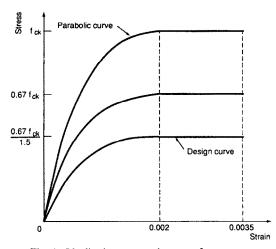
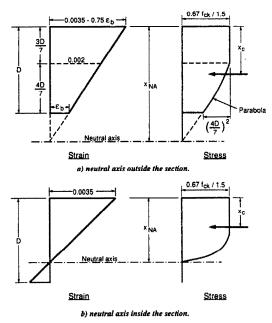


Fig. 1. Idealized stress-strain curve for concrete.

here apply to relatively short columns for which the effect of lateral deflections on the magnitude of bending moments is negligible. Also effects of sustained load and reversal of bending moments are not considered.

In the present investigation, a square section of size  $B \times B$ is considered. If a small square of size  $B_1 \times B_1 (B_1 < B)$  is removed from the corner of a original square section then it will become a symmetric L-section as shown in Fig. 5. For different values of  $B_1$  the L-section so f various sizes can be obtained. If  $B_1 = 0.0$ , the L-section becomes square section. Since the ratio of  $B_1/B$  for all practical purposes varies from 0.3 to 0.6, the present work is limited to the ratios of  $B_1/B$ equal to 0.3, 0.4, 0.5 and 0.6. The parameters considered are symmetric L-shaped column sections and reinforcement is assumed to be uniformly distributed as a thin strip along all the sides with effective cover to depth ratio (B'/B) as 0.1.

Design charts for combined axial compression and bending are obtained in the form of interaction diagrams in which curves for  $P_u/f_{ck}B^2$  versus  $M_u/f_{ck}B^3$  are plotted for different values of  $p/f_{ck}$ . When bending moments are acting



x<sub>NA</sub> = depth of neutral axis

Fig. 2. Stress and strain diagrams.

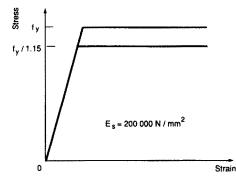


Fig. 3. Idealized stress-strain curve for mild steel bars.

in addition to axial load, the points for plotting the interaction diagrams are obtained by assuming different positions of neutral axis. For each position of the neutral axis, the strain distribution across the section and the stress block parameters are determined. The stresses in the reinforcement are also calculated from the known strains. Thereafter the resultant axial force and the moment about the centroid of the section are calculated as follows.

To find the forces and moments due to concrete in the L-section subjected to axial compression and bending (both uniaxial and biaxial bending with equal eccentricities  $e_x = e_y = e$ ), the following procedure is used in the analysis. The stress block (see Fig. 2) is divided into number of strips. First the width of each strip is calculated. This strip width is multiplied by corresponding width of the section and depth of the strip, which gives the force in that strip of concrete. The algebraic sum of all such elemental forces gives the total force in concrete. This force in concrete multiplied by the distance between centroid of the stress block and centroid of the section gives the moment due to concrete. The forces and moments due to reinforcement (both for uniaxial and biaxial bending) are determined as follows:

force in the reinforcement = 
$$\sum_{i=1}^{n} (f_{si} - f_{ci}) p_i A_c / 100$$
 (1a)

moment of resistance

with respect to steel = 
$$\sum_{i=1}^{n} (f_{si} - f_{ci})p_i A_c y_i / 100$$
 (1b)

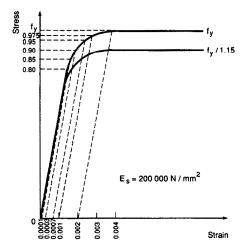
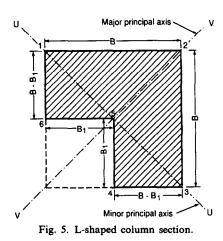


Fig. 4. Idealized stress-strain curve for high yield strength deformed bars.



in which *n* is the number of rows of reinforcement, but in the present work, since the reinforcement is assumed to be distributed uniformly as thin strip along all sides of the L-section, the notation *n* here refers to the number of unit length of the reinforcement strip at which the stresses in steel and concrete (at that level) are to be determined,  $f_{si}$  is the stress in the *i*th row of steel (compression being positive and tension negative),  $f_{ci}$  is the stress in concrete at the level of the *i*th row of reinforcement,  $A_c$  is the area of concrete, may be taken equal to the gross area,  $p_i = (A_{si}/A_c)100$  is the percentage of steel in the *i*th row,  $A_{si}$  is the area of reinforcement in the *i*th row,  $y_i$  is the distance of the *i*th row of reinforcement measured from the centroid of the section. It is positive towards the highly compressed edge and negative towards the least compressed edge.

#### INTERACTION DIAGRAMS

Because of symmetry, in a square section, the eccentricity on either side of the centre of gravity makes no difference in the approach to interaction curves except when steel is not symmetric, whereas, in the case of L-shaped column sections the same is not true. For the L-shaped section considered here, the minor principal axis U-U and major principal axis V-V are shown in Fig. 5, and the interaction curves have been prepared by considering the axis of bending as explained below:

- Case 1. Uniaxial bending parallel to edge 1-2, by treating edge 1-2 in compression.
- Case 2. Uniaxial bending parallel to edge 1-2, by treating edge 1-2 in tension.
- Case 3. Biaxial bending with equal eccentricities treating corner 2 in compression.
- Case 4. Biaxial bending with equal eccentricities treating corner 2 in tension.

The four computer programs, namely UNIAX1, UNIAX2, BIAX1, and BIAX2 for the above-mentioned cases 1-4, respectively, are developed by using FORTRAN and are presented in the Appendix. These programs were used to obtain the ultimate bload  $(P_u)$  and moment  $(M_u)$  as

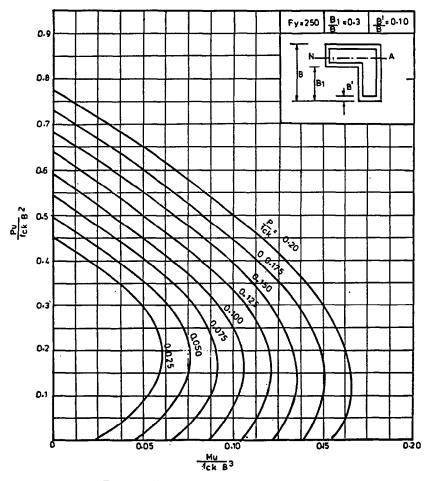
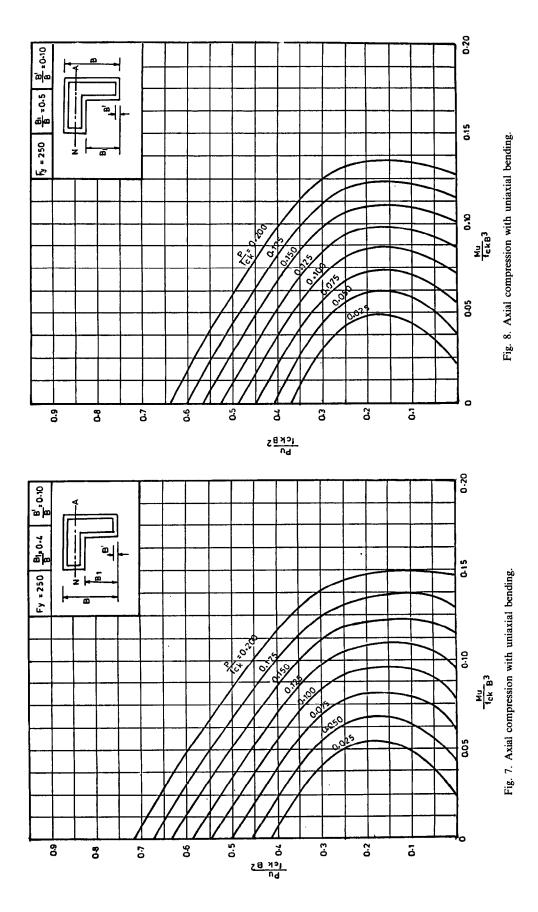
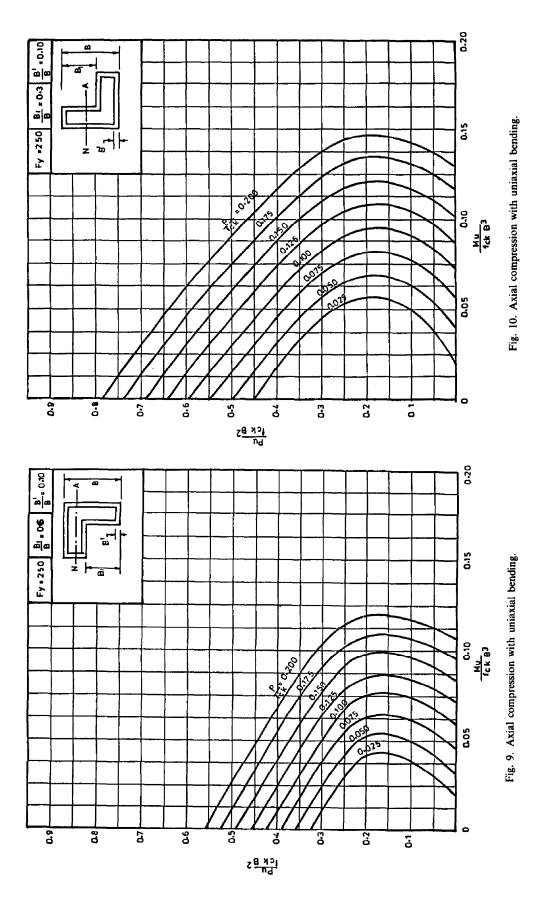


Fig. 6. Axial compression with uniaxial bending.







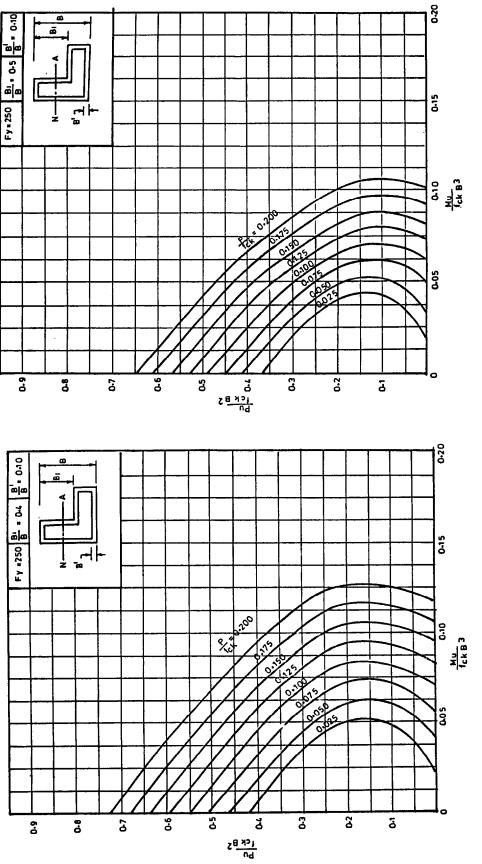
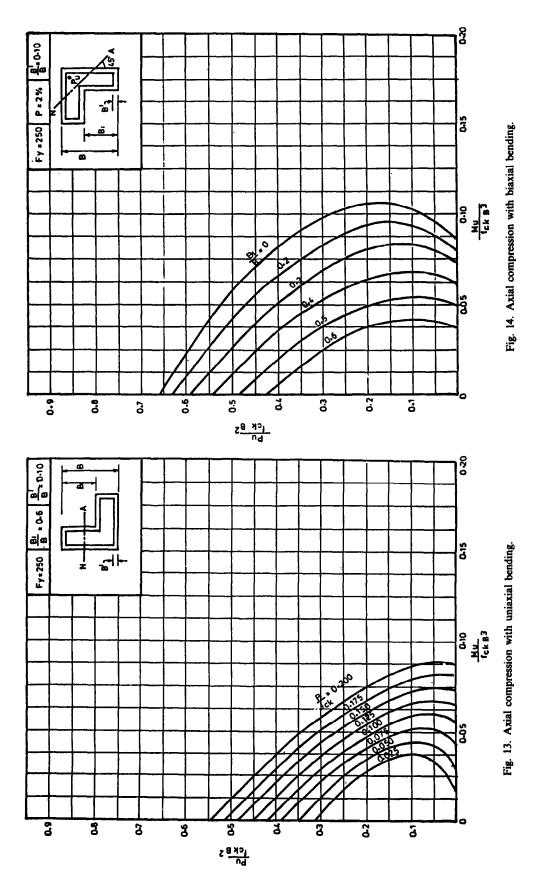
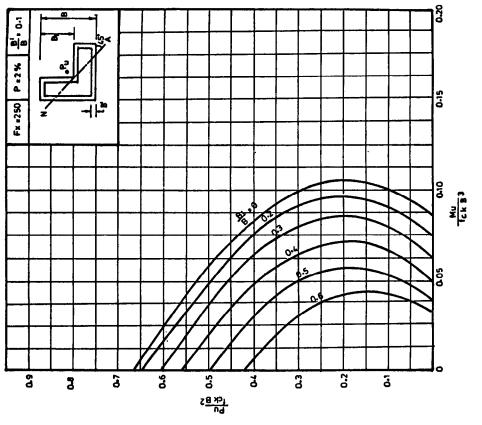
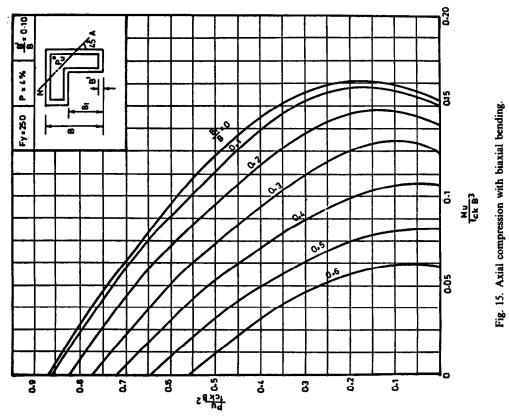


Fig. 12. Axial compression with uniaxial bending.

Fig. 11. Axial compression with uniaxial bending.







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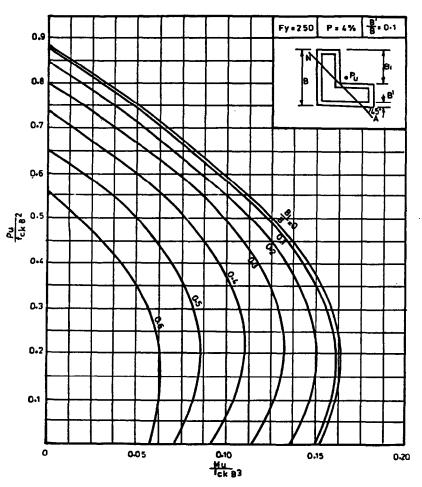


Fig. 17. Axial compression with biaxial bending.

output for different positions of neutral axes. The input data consists of the square size  $(B_1)$  of the removed portion from the original square section, depth (B) of the section, cover depth (B'), characteristic strength of the concrete  $(f_{ck})$  and steel  $(f_y)$ , and modulus of elasticity of steel  $(E_c)$ . All the loads and moments so obtained are graphically represented in terms of non-dimensional parameters  $(P_u/f_{ck}B^2)$  versus  $M_u/f_{ck}B^3$  and are shown in Figs 6-17.

## LIMITATIONS

The number of variables considered in this paper are restricted as there was limited space. In the present investigation, the following parameters were considered:

- (i) symmetric L-shaped column sections with reinforcement as a thin strip along all sides,
- (ii) effective cover to depth ratio (B'/B) is taken as 0.1 in all the cases,
- (iii)  $B_1/B$  ratios are 0.0, 0.3, 0.4, 0.5 and 0.6,
- (iv) the modulus of elasticity of mild steel is taken equal to 200 kN/mm<sup>2</sup>.

### CONCLUSIONS

The analysis of reinforced concrete L-shaped column sections subjected to axial compression and bending (uniaxial and biaxial) has been computerized. Interaction curves for L-shaped column sections under axial compression and uniaxial bending for two cases are presented in Figs 6-9 and Figs 10-13. For columns under axial compression and biaxial bending with equal eccentricities, the curves for two cases are shown in Figs 14-15 and Figs 16-17. It is hoped that the charts which are included in this paper, will be useful aids for designers and also will bring some attention to the particular form of resistance exhibited by these cross-sections. It offers the possibility of economizing and can complement the existing design procedures.

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### APPENDIX

C1	*****	***************************************
C C		ANALYSIS OF L-SHAPED COLUMN SECTIONS UNDER AXIAL COMPRESSION AND UNIAXIAL BENDING (See Figs. 6-9)
C,		
		OPEN(5,FILE='UNIAX1.DAT') OPEN(6,FILE='UNIAX1.OUT',STATUS='NEW')
		DO 299 I=1,10
		READ (5, 5, END=300) B1, B, DC, FCK, FY, ES
	5	FORMAT (5F6.2, F10.2)
	•	P=0.0
	7	P=P+0.5
	•	XU=DC
	8	XU=XU+5.0
	Ŧ	FK=0.446*FCK
		Y1 = (0.5 * B* (B-B1) + B1* (B-0.5 * B1)) / (B+B1)
		A=B*(B-B1)+B1*(B-B1)
		AS=P*A/100.0
		RL=4.0*B-8.0*DC
		TS=AS/RL
С		DETERMINATION OF FORCES DUE TO CONCRETE
		IF(XU-B) 10,10,20
	10	X1=3.0*XU/7.0
		X2=4.0*XU/7.0
	~ ~	GO TO 30
	20	X1=3.0*B/7.0
	20	X2=4.0*B/7.0
	30	PC=0.0 AMC=0.0
		B5= (B-B1)
		IF(X1-B5) 40,70,70
	40	C1=B*X1*FK
	10	BMC1=C1*(Y1-0.5*X1)
		X=0.0
	45	x=x+1.0
		B6=B-B1-X1
		IF (X-B6) 50, 50, 60
	50	F=FK-FK*X*X/((XU-X1)*(XU-X1))
		PC1=B*F
		AMC1=PC1*(Y1-(X1+X))
		GO TO 65
	60	F=FK-FK*X*X/((XU-X1)*(XU-X1))
		PC1=(B-B1)*F
	~-	AMC1=PC1*(Y1-(X1+X))
	65	PC=PC+PC1
		AMC=AMC+AMC1
		IF(X.LT.X2) GO TO 45 TPC=PC+C1

		TAMC=AMC+BMC1
	70	GO TO 80
	70	C1=(B-B1)*B*FK BMC1=C1*(Y1-0.5*(B-B1))
		C2 = (B-B1) * (X1 - (B-B1)) * FK
		AMC2=C2*(Y1-(0.5*X1-0.5*(B-B1)))
		X=0.0
	75	X=X+1.0
		F=FK-FK*X*X/((XU-X1)*(XU-X1))
		PC1=(B-B1)*F AMC1=PC1*(Y1-(X1+X))
		PC=PC+PC1
		AMC=AMC+AMC1
		IF (X.LT.X2) GO TO 75
		TPC=C1+C2+PC
с		TAMC=BMC1+AMC2+AMC DETERMINATION OF FORCES IN COMPRESSION
č		REINFORCEMENT
	80	B7=B-DC
		IF(XU-B7) 90,160,160
	90	PSC=0.0
		AMSC=0.0 PSC1=0.0
		AMSC1=0.0
		D=0.0
	95	D=D+1.0
		EC=0.0035*D/XU
		IF(FY.EQ.250.0)GO TO 96 IF(EC.GE.0.0038) GO TO 96
		IF (EC.LE.0.00145) GO TO 96
		FC=FY/1.15-12831145.0*(0.0038-EC)*(0.0038-EC)
		GO TO 97
	96	FC=EC*ES
		F1=FY/1.15 IF(FC.GE.F1) FC=F1
	97	B8=XU-(B-B1-DC)
		IF(D-B8) 110,100,110
	100	PSC1=B1*TS* (FC-FK)
		AMSC1=PSC1*(Y1-(XU-D)) GO TO 115
	110	PSC2=2.0*TS* (FC-FK)
		AMSC2=PSC2*(Y1-(XU-D))
		PSC=PSC+PSC2
		AMSC=AMSC+AMSC2
	115	B9=XU-DC IF(D.LT.B9) GO TO 95
	110	PSC3 = (B-2.0*DC) *TS* (FC-FK)
		AMSC3=PSC3* (Y1- (XU-D))
		TPSC=PSC+PSC1+PSC3
~		TAMSC=AMSC+AMSC1+AMSC3
С		DETERMINATION OF FORCES DUE TO TENSILE REINFORCEMENT Z1=(B-DC-XU)
		PST=0.0
		AMST=0.0
		PST1=0.0
		AMST1=0.0
	116	Z=0.0 Z=Z+1.0
	110	$ET=0.0035 \times Z / XU$
		IF (FY.EQ.250.0) GO TO 117
		IF(ET.GE.0.0038) GO TO 117
		IF (ET.LE.0.00145) GO TO 117
		FT=FY/1.15-12831145.0*(0.0038-ET)*(0.0038-ET) GO TO 118
	117	FT=ET*ES
		F1=FY/1.15
		IF (FT.GE.F1) FT=F1
	118	22=(B-B1-DC-XU) IF(Z-Z2) 120,130,120
		IE (4-42) IZU, ISU, IZU

130	PST1=-B1*TS*FT
	AMST1=PST1*(Y1-(XU+Z))
	GO TO 125
120	PST2=-2.0*TS*FT
	AMST2=PST2*(Y1-(XU+Z))
	PST=PST+PST2
105	AMST=AMST+AMST2
125	IF(Z.LT.Z1) GO TO 116
	PST3=- (B-2.0*DC) *TS*FT
	AMST3=PST3*(Y1-(XU+Z))
	TPST=PST+PST1+PST3
	TAMST=AMST+AMST1+AMST3
	GO TO 200
160	FT=0.0
	PSC=0.0
	AMSC=0.0
	PSC2=0.0
	AMSC2=0.0
	D1=(XU-(B-DC))
	D=D1-1.0
165	D=D+1.0
	EC=0.002*D/(XU-X1)
	IF (FY.EQ.250.0) GO TO 168
	IF (EC.GE.0.0038) GO TO 168
	IF (EC.LE.0.00145) GO TO 168
	FC=FY/1.15-12831145.0*(0.0038-EC)*(0.0038-EC)
	GO TO 169
168	FC=EC*ES
	F1=FY/1.15
	IF(FC.GE.F1)FC=F1
169	IF(D-D1) 166,167,166
167	PSC3 = (B-B1-2.0*DC) *TS* (FC-FK)
	AMSC3=PSC3*(Y1-(XU-D))
	GO TO 165
166	B11=XU- (B-B1-DC)
	IF(D-B11) 170,180,170
180	PSC2=B1*TS* (FC-FK)
	AMSC2=PSC2*(Y1-(XU-D))
	GO TO 175
170	
170	PSC4=2.0*TS*(FC-FK)
	AMSC4=PSC4*(Y1-(XU-D))
	PSC=PSC+PSC4
	AMSC=AMSC+AMSC4
	B12=XU-DC
175	IF(D.LT.B12) GO TO 165
	PSC1 = (B-2.0*DC) *TS* (FC-FK)
	AMSC1=PSC1*(Y1-(XU-D))
	TPSC=PSC+PSC1+PSC2+PSC3
	TAMSC=AMSC+AMSC1+AMSC2+AMSC3
	TPST=0.0
	TAMST=0.0
200	TF=TPSC+TPST+TPC
	TM=TAMSC+TAMST+TAMC
	E=TM/TF
	XC=TM/ (FCK*B*B*B)
	YC=TF/(FCK*B*B)
	P1=P/FCK
	WRITE(6,6) XU,P1,FT,XC,TC
6	FORMAT (5X, 'XU=', F5.1, 5X, 'P1=', F5.4, 5X, 'FT=', F5.1,
Ū	5x, 'xC=', F5.4, 5x, 'YC=', F5.4/)
	XUMAX=2.0*B
	IF (XU.LE.XUMAX) GO TO 8
	IF (P.LT.4.0) GO TO 7
299	CONTINUE
300	STOP
200	
<u>^</u>	END

## Technical Note

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c c	P C	ANALYSIS OF L-SHAPED COLUMN SECTIONS UNDER AXIAL COMPRESSION AND UNIAXIAL BENDING (See Figs. 10-13)
		OPEN(5,FILE='UNIAX2.DAT') OPEN(6,FILE='UNIAX2.OUT',STATUS='NEW') DO 299 I=1,10
	5	READ (5, 5, END=300) B1, B, DC, FCK, FY, ES FORMAT (5F6.2, F10.2)
	7	P=0.0 P=P+0.5
	8	XU=DC XU=XU+5.0
	•	FK=0.446*FCK A=B*(B-B1)+B1*(B-B1)
		Y1=B-(0.5*B*(B-B1)+B1*(B-0.5*B1))/(B+B1) AS=P*A/100.0
		RL=4.0*B-8.0*DC
с		TS=AS/RL DETERMINATION OF FORCES DUE TO CONCRETE
	10	IF(XU-B) 10,10,20 X1=3.0*XU/7.0
		X2=XU-X1 GO TO 30
	20	X1=3.0*B/7.0 X2=B-X1
	30	TPC=0.0 TAMC=0.0
	35	X=0.0 X=X+1.0
	40	IF(X-X1)40,40,50 F=FK
	50	$ \begin{array}{c} F = F \\ GO & TO \\ F = F \\ K - F \\ K \\ (X - X1) $
	60	IF(X-B1) 70,100,100
	70	W=B-B1 GO TO 130
	100 130	W=B PC=W*F
		AMC=PC*(Y1-X) TPC=TPC+PC
		TAMC=TAMC+AMC B5=X1+X2
с		IF(X.LT.B5) GO TO 35 DETERMINATION OF FORCES IN COMPRESSION REINFORCEMENT
-		TPSC=0.0 TAMSC=0.0
		B6=B-DC IF (XU~B6) 140, 150, 150
	140	D=0.0
	150	GO TO 160 D=XU-B6-1.0
	160	D=D+1.0 EC=0.002*D/(XU-X1)
		IF(FY.EQ.250.0) GO TO 168 IF(EC.GE.0.0038) GO TO 168
		IF(EC.LE.0.00145) GO TO 168 FC=FY/1.15-12831145.0*(0.0038-EC)*(0.0038-EC)
	168	GO TO 169 FC=EC*ES
		F1=FY/1.15 IF(FC.GE.F1) FC=F1
	169	B7=XU-B+DC B8=XU-B1-DC
		B11=XU-DC IF(D.EQ.B7) GO TO 180
		IF (D.EQ.B1) GO TO 100 IF (D.EQ.B11) GO TO 200 IF (D.EQ.B8) GO TO 190
		WS=2.0*TS

		GO TO 210
	180	WS = (B-2.0*DC)*TS
		GO TO 210
	190	WS=B1*TS
		GO TO 210
	200	WS = (B - B1 - 2.0 * DC) * TS
	210	PSC=WS* (FC-FK)
		AMSC=PSC*(Y1-(XU-D))
		TPSC=TPSC+PSC
		TAMSC=TAMSC+AMSC
		IF (D.LT.B11) GO TO 160
с		DETERMINATION OF FORCES IN TENSION STEEL
v		FT=0.0
		TPST=0.0
		TAMST=0.0
		IF (XU.GE.B6) GO TO 265
		2=0.0
	215	Z=Z+1.0
	215	ET=0.002*Z/(XU-X1)
		IF (FY.EQ.250.0) GO TO 216
		IF (FI.EQ.230.0) GO TO 216
		IF (ET.LE.0.00145) GO TO 216
		FT = FY/1.15 - 12831145.0 * (0.0038 - ET) * (0.0038 - ET)
	216	GO TO 217 FT=ET*ES
	210	
		F1=FY/1.15
	017	IF(FT.GE.F1) FT=F1
	217	B12=B1+DC-XU
		B14=B-DC-XU
		IF(Z.EQ.B12) GO TO 230
		IF(Z.EQ.B14) GO TO 240
		WS=2.0*TS
	000	GO TO 260
	230	WS=B1*TS
	240	GO TO 260
	240	$WS = (B-2.0 \times DC) \times TS$
	260	PST=-WS*FT
		AMST=PST*(Y1-(XU+Z))
		TPST=TPST+PST
		TPST=TPST+PST TAMST=TAMST+AMST
	0.65	TPST=TPST+PST TAMST=TAMST+AMST IF(Z.LT.B14) GO TO 215
	265	TPST=TPST+PST TAMST=TAMST+AMST IF(Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST
	265	TPST=TPST+PST TAMST=TAMST+AMST IF(Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST
	265	TPST=TPST+PST TAMST=TAMST+AMST IF(Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF
	265	TPST=TPST+PST TAMST=TAMST+AMST IF(Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK
	265	TPST=TPST+PST TAMST=TAMST+AMST IF(Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B*B)
	265	TPST=TPST+PST TAMST=TAMST+AMST IF(Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B*B) YC=TF/(FCK*B*B)
		TPST=TPST+PST TAMST=TAMST+AMST IF(Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B*B) YC=TF/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC
	265	TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/ (FCK*B*B) YC=TF/ (FCK*B*B) WRITE (6, 6) XU,P1,FT,XC,TC FORMAT (5X, 'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1,
		TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/ (FCK*B*B) YC=TF/ (FCK*B*B) WRITE (6,6) XU,P1,FT,XC,TC FORMAT (5X, 'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/)
		TPST=TPST+PST TAMST=TAMST+AMST IF(2.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC FORMAT(5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B
		TPST=TPST+PST TAMST=TAMST+AMST IF(2.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC FORMAT(5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF(XU.LT.XUMAX) GO TO 8
	6	TPST=TPST+PST TAMST=TAMST+AMST IF(Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC FORMAT(5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF(XU.LT.XUMAX) GO TO 8 IF(P.LT.4.0) GO TO 7
	6 299	TPST=TPST+PST TAMST=TAMST+AMST IF(2.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC FORMAT(5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF(XU.LT.XUMAX) GO TO 8 IF(P.LT.4.0) GO TO 7 CONTINUE
	6	TPST=TPST+PST TAMST=TAMST+AMST IF(2.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC FORMAT(5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF(XU.LT.XUMAX) GO TO 8 IF(P.LT.4.0) GO TO 7 CONTINUE STOP
	6 299 300	TPST=TPST+PST TAMST=TAMST+AMST IF(2.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC FORMAT(5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF(XU.LT.XUMAX) GO TO 8 IF(P.LT.4.0) GO TO 7 CONTINUE
Ċ	6 299 300	TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/ (FCK*B*B) WRITE (6,6) XU,P1,FT,XC,TC FORMAT (5X, 'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X, 'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END
Č	6 299 300	TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC FORMAT(5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END
Č' C	6 299 300	TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/ (FCK*B*B) WRITE (6,6) XU,P1,FT,XC,TC FORMAT (5X, 'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X, 'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END ************************************
C C C	6 299 300	TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE (6,6) XU,P1,FT,XC,TC FORMAT (5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END ************************************
0 0 0 0 0	6 299 300	TPST=TPST+PST TAMST=TAMST+AMST IF (2.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE (6,6) XU,P1,FT,XC,TC FORMAT (5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END ************************************
0 0 0 0 0	6 299 300	TPST=TPST+PST TAMST=TAMST+AMST IF (2.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC FORMAT(5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END ************************************
0 0 0 0 0	6 299 300	TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/ (FCK*B*B) WRITE (6,6) XU,P1,FT,XC,TC FORMAT (5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END ANALYSIS OF L-SHAPED COLUMN SECTIONS SUBJECTED TO AXIAL COMPRESSION AND BIAXIAL BENDING WITH EQUAL ECCENTRICITIES (See Figs.14 and 15)
0 0 0 0 0	6 299 300	TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC FORMAT(5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END ************************************
0 0 0 0 0	6 299 300	TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC FORMAT(5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END ************************************
0 0 0 0 0	6 299 300 ******	TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/ (FCK*B*B) WRITE (6,6) XU,P1,FT,XC,TC FORMAT (5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END ************************************
0 0 0 0 0	6 299 300	TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/(FCK*B*B) WRITE(6,6) XU,P1,FT,XC,TC FORMAT(5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END ************************************
0 0 0 0 0	6 299 300 ******	TPST=TPST+PST TAMST=TAMST+AMST IF (Z.LT.B14) GO TO 215 TF=TPC+TPSC+TPST TM=TAMC+TAMSC+TAMST E=TM/TF P1=P/FCK XC=TM/ (FCK*B*B) WRITE (6,6) XU,P1,FT,XC,TC FORMAT (5X,'XU=',F6.1,5X,'P1=',F5.4,5X,'FT=',F5.1, 5X,'XC=',F6.4,5X,'YC=',F6.4/) XUMAX=2.0*B IF (XU.LT.XUMAX) GO TO 8 IF (P.LT.4.0) GO TO 7 CONTINUE STOP END ************************************

	8	XU=DC XU=XU+5.0
	0	FK=0.446*FCK
		A=B*(B-B1)+B1*(B-B1)
		Y1= (B*B*B*0.707-B1*B1*B1*0.707-B1*B1*(B-B1)*1.414)/A AS=P*A/100.0
		RL=4.0*B-8.0*DC
~		TS=AS/RL
С		DETERMINATION OF FORCES DUE TO CONCRETE B2=(2.0*B-B1)*0.707
		IF (XU-B2) 10,10,20
	10	X1=3.0*XU/7.0
		X2=XU-X1 GO TO 30
	20	X1=3.0*B2/7.0
	~ ~	X2=B2-X1
	30	TPC=0.0 TAMC=0.0
		X=0.0
	35	X=X+1.0
	40	IF(X-X1)40,40,50 F=FK
	40	GO TO 60
	50	F=FK-FK*(X-X1)*(X-X1)/(XU-X1)/(XU-X1)
	60	B3=1.4142*(B-B1) B4=0.707*B
		IF (X-B3) 70,70,100
	70	IF(X-B4) 80,80,90
	80	W=2.0*X GO TO 130
	90	W=2.0*(1.4142*B-X)
		GO TO 130
	100 110	IF(X-B4) 110,110,120 W=2.0*1.4142*(B-B1)
	110	GO TO 130
	120	$W = (2.0 \times B - B1 - X \times 1.4142) \times 2.0 \times 1.4142$
	130	PC=W*F AMC=PC*(Y1-X)
		TPC=TPC+PC
		TAMC=TAMC+AMC
		B5=X1+X2 IF(X.LT.B5) GO TO 35
с		DETERMINATION OF FORCES IN COMPRESSION REINFORCEMENT
-		TPSC=0.0
		TAMSC=0.0
		B6= (2.0*B-B1-2.0*DC)*0.7071 IF (XU-B6)140,150,150
	140	D=0.0
	1 5 0	GO TO 160
	150 160	D=XU~B6-1.0 D=D+1.0
		EC=0.002*D/(XU-X1)
		IF (FY.EQ.250.0) GO TO 168
		IF(EC.GE.0.0038) GO TO 168 IF(EC.LE.0.00145) GO TO 168
		FC=FY/1.15-12831145.0*(0.0038-EC)*(0.0038-EC)
	168	GO TO 169 FC=EC*ES
	100	F1=FY/1.15
		IF(FC.GE.F1) FC=F1
	169	B7=B*0.7071
		B8= (B-B1-DC) *1.4142 B9=XU-B7
		B10=XU-B8
		B11=XU-1.4142*DC IF(D~B10) 180,180,190
	180	WS=4.0*TS*1.4142
		GO TO 210

	190	WS=2.0*TS*1.414
	210	PSC=WS*(FC-FK)
		AMSC=PSC*(Y1-(XU-D))
		TPSC=TPSC+PSC
		TAMSC=TAMSC+AMSC
с		IF(D.LT.B11) GO TO 160 DETERMINATION OF FORCES IN TENSION STEEL
C		FT=0.0
		TPST=0.0
		TAMST=0.0
		IF (XU.GE.B6) GO TO 265
		Z=0.0
	215	Z=Z+1.0
		ET=0.002*Z/(XU-X1)
		IF (FY.EQ.250.0) GO TO 216
		IF (ET.GE.0.0038) GO TO 216
		IF (ET.LE.0.00145) GO TO 216
		FT=FY/1.15-12831145.0*(0.0038-ET)*(0.0038-ET)
		GO TO 217
	216	FT=ET*ES
		F1=FY/1.15
		IF(FT.GE.F1) FT=F1
	217	B12=-B9
		B13=-B10 B14=B6-XU
		IF(Z-B13) 230,230,240
	230	WS=2.0*TS*1.4142
	200	GO TO 260
	240	WS=4.0*TS*1.4142
	260	PST=-WS*FT
	200	AMST=PST* (Y1- (XU+Z))
		TPST=TPST+PST
		TAMST=TAMST+AMST
		IF(Z.LT.B14) GO TO 215
	265	TF=TPC+TPSC+TPST
		TM=TAMC+TAMSC+TAMST
		E=TM/TF
		P1=P/FCK
		XC=TM/ (FCK*B*B*B)
		YC=TF/(FCK*B*B) WRITE(6,6) XU,P1,XC,TC
	6	FORMAT (5x, 'XU=', F7.1, 5x, 'P1=', F8.5,
	0	5X, 'XC=', F6.4, 5X, 'YC=', F6.4/)
		XUMAX=1.5*B
		IF (XU.LT.XUMAX) GO TO 8
		IF (P.LT.4.0) GO TO 7
	299	CONTINUE
	300	STOP
		END
C		
С		*****
С		ALYSIS OF L-SHAPED COLUMN SECTIONS SUBJECTED TO
С		IAL COMPRESSION AND BIAXIAL BENDING WITH EQUAL
c	EC	CENTRICITES (See Figs.16 and 17)
C	*****	OPEN (5, FILE='BIAX2.DAT')
		OPEN (5, FILE=' BIAX2.DAT') OPEN (6, FILE=' BIAX2.OUT', STATUS=' NEW')
		DO 299 I=1,10
		READ (5, 5, END=300) B1, B, DC, FCK, FY, ES
	5	FORMAT (5F6.2, F10.2)
		P=0.0
	7	P=P+0.5
	_	XU=DC
	8	XU=XU+5.0
		FK=0.446*FCK
		A=B*(B-B1)+B1*(B-B1) Y1=(B*B*B*0.707-B1*B1*B1*0.707-B1*B1*(B-B1)*1.414)/A
		$M = (B \times B \times B \times 0.707 - B1 \times B1 \times B1 \times 0.707 - B1 \times B1 \times (B - B1) \times 1.41477 A$ AS=P*A/100.0

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с		RL=4.0*B-8.0*DC TS=AS/RL DETERMINATION OF FORCES DUE TO CONCRETE B2=(2.0*B-B1)*0.707 Y2=B2-Y1
	10	IF (XU-B2) 10,10,20 X1=3.0*XU/7.0 X2=XU-X1 GO TO 30
	20	X1=3.0*B2/7.0
	30	X2=B2-X1 TPC=0.0 TAMC=0.0
	35	x=0.0 x=x+1.0
	40	IF (X-X1) 40, 40, 50 F=FK
	50	GO TO 60 $F = FY + FY + (Y + Y1) + (Y + Y1) + (Y + Y1) + (Y + Y1)$
	50 60	F=FK-FK*(X-X1)*(X-X1)/(XU-X1)/(XU-X1) B3=1.4142*(B-B1)
	00	B4=0.707*B
		B31=B2-B3
		B41=B2-B4
		IF(X-B31) 70,70,100
	70	IF(X-B41) 80,80,90
	80	W=4.0*X GO TO 130
	90	W = (B-B1) * 1.4142 * 2.0
		GO TO 130
	100	IF(X-B41) 110,110,120
	110	W = (1.4142 * B - B2 + X) * 2.0
	120	GO TO 130 W=(B2-X)*2.0
	130	PC=W*F
	100	AMC=PC* (Y2-X)
		TPC=TPC+PC
		TAMC=TAMC+AMC
		B5=X1+X2
~		IF (X.LT.B5) GO TO 35
С		DETERMINATION OF FORCES IN COMPRESSION REINFORCEMENT TPSC=0.0
		TAMSC=0.0
		B6= (2.0*B-B1-2.0*DC) *0.7071
		IF (XU-B6) 140, 150, 150
	140	D=0.0
	. 1 5 0	GO TO 160
	150 160	D=XU-B6-1.0 D=D+1.0
	100	EC=0.002*D/(XU-X1)
		IF (FY.EQ.250.0) GO TO 168
		IF(EC.GE.0.0038) GO TO 168
		IF (EC.LE.0.00145) GO TO 168
		FC=FY/1.15-12831145.0*(0.0038-EC)*(0.0038-EC) GO TO 169
	168	FC=EC*ES
		F1=FY/1.15
		IF(FC.GE.F1) FC=F1
	169	B7=B*0.7071
		B8=(B-B1-DC)*1.4142 B9=XU-B7
		BJ0≈XU-B2+B8
		B11≈XU-1.4142*DC
		IF(D-B10) 180,180,190
	180	WS=2.0*TS*1.4142
	190	GO TO 210 WS=4.0*TS*1.414
	210	WS=4.0*TS*1.414 PSC=WS* (FC-FK)
		AMSC=PSC* (Y2~ (XU-D))
		TPSC=TPSC+PSC

		TAMSC=TAMSC+AMSC
		IF(D.LT.B11) GO TO 160
С		DETERMINATION OF FORCES IN TENSION STEEL
		FT=0.0
		TPST=0.0
		TAMST=0.0
		IF (XU.GE.B6) GO TO 265
		Z=0.0
	215	Z = Z + 1, 0
		ET=0.002*Z/(XU-X1)
		IF (FY.EQ.250.0) GO TO 216
		IF (ET.GE.0.0038) GO TO 216
		IF (ET.LE.0.00145) GO TO 216
		FT=FY/1.15-12831145.0*(0.0038-ET)*(0.0038-ET)
		GO TO 217
	216	FT=ET*ES
		F1=FY/1.15
		IF(FT.GE.F1) FT=F1
	217	B12=-B9
		B13=B2-B8-XU
		B14=B6-XU
		IF(Z-B13) 230,230,240
	230	WS=4.0*TS*1.4142
		GO TO 260
	240	WS=2.0*TS*1.4142
	260	PST=-WS*FT
		AMST=PST*(Y2-(XU+Z))
		TPST=TPST+PST
		TAMST=TAMST+AMST
		IF(Z.LT.B14) GO TO 215
	265	TF=TPC+TPSC+TPST
		TM=TAMC+TAMSC+TAMST
		E=TM/TF
		P1=P/FCK
		XC=TM/(FCK*B*B*B)
		YC=TF/(FCK*B*B)
	~	WRITE(6,6) XU,P1,XC,YC
	6	FORMAT (5X, 'XU=', F7.1, 5X, 'P1=', F8.5,
		5X, 'XC=', F6.4, 5X, 'YC=', F6.4/)
		XUMAX=1.5*B
		IF (XU.LT.XUMAX) GO TO 8
	200	IF(P.LT.2.0) GO TO 7
	299	CONTINUE
	300	STOP
<b>C</b>		END
_C-		